



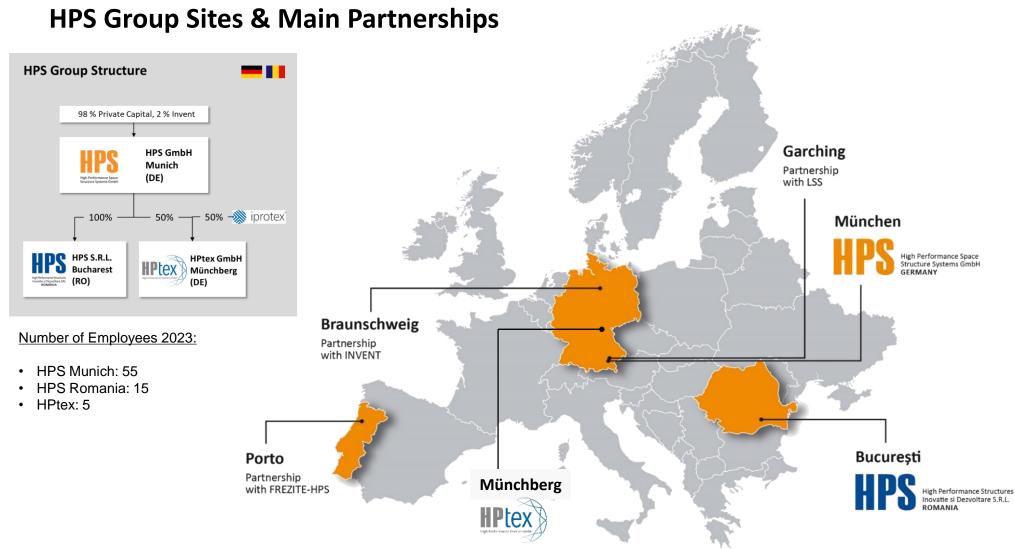
The ADEO Space Sail Products

Daniel Stelzl, Patric Seefeldt (DLR), Matthias Killian et al.

ADEO Team (HPS)

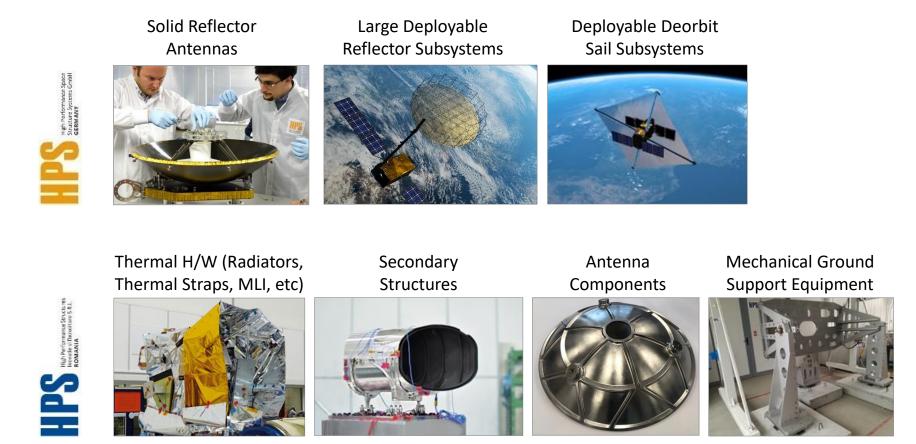
Munich | 2023







Product Portfolio HPS Group



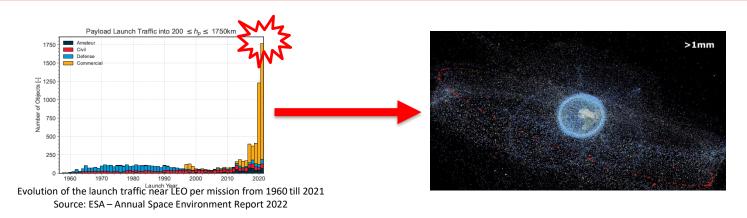
Other Products and Services: radiation protection caps, composite structures, engineering & integration service



Space Debris – Deorbiting – Current & New Regulations



Past: 25 year rule/recommendation from IADC was derived before the "NewSpace" age and does NOT reflect the current situation of launches and S/C in orbit



Current: New FCC-Regulation (USA), Zero Debris Policy (ESA) and Space Traffic Management (EC):

Decommission all LEO S/C out of at the end of their operational live

within **≤ 5 YEARS**

with **DEORBIT RELIABLITY** $\rightarrow \geq 90\%$



Contributor to several ESA, European and National – Debris Avoidance and CleanSpace Commites and Associations:



CleanSpace Initiative & Zero Debris Policy



ESOC Space Safety Programs



European Space Traffic Management

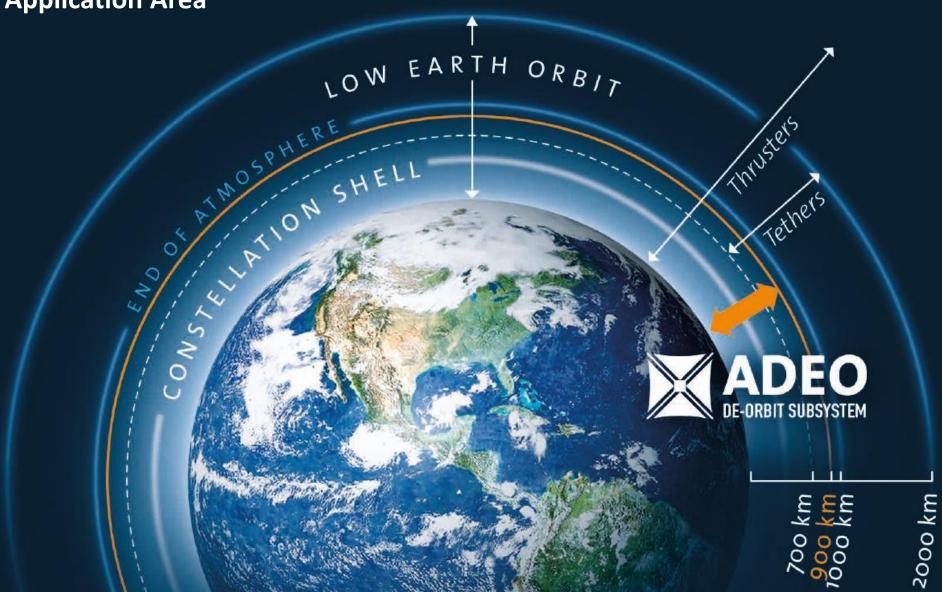


German BDI NewSpace Initiative – sustainable astronautics



ADEO - THE DEORBIT MODULE FROM LEO

Application Area





Generic Active vs. Passive Deorbiting Comparison



Active Deorbiting (e.g. propulsion systems)

- Active deorbiting system need an active S/C and propulsion system but they are failure prone
- Active deorbiting require high operational expenses required (OPEX)
- ADR Active Debris Removal very expensive

Passive (automatic) Deorbiting (e.g. dragsails or tethers)

- Prolonging missions due to autonomous capability (ADEO)
- No active S/C necessary \rightarrow Fail Safe with autonomous function (ADEO)
- Solution for S/C with no propulsion system e.g. CubeSats, Rocket Bodies
- Low/No operational expenses



From Science to Product \Rightarrow





Over the last 10 years different space sail technologies have been industrialize

originating from DLR research and development projects going back at least to 1990s.



M. Leipold, C.E. Garner, R. Freeland, et al. ODISSEE, A proposal for demonstration of a solar sail in earth orbit, Acta Astronautica 1999 45/4, 557-566 D. Agnolon, Study overview of a solar sail demonstrator: GEOSAIL, DLR/ESA, 2008 N. Wolff, P. Seefeldt, W. Bauer, et al., Alternative application of solar sail technology. In Advances in Solar Sailing, pages 351-365. Springer, 2014. P. Seefeldt, P. Spiez, T. Spröwitz, et al., Gossamer-1: Mission Concept and Technology for a Controlled Deployment of Gossamer Spacecraft, Advances in Space Research 59.1 (2017): 434-456

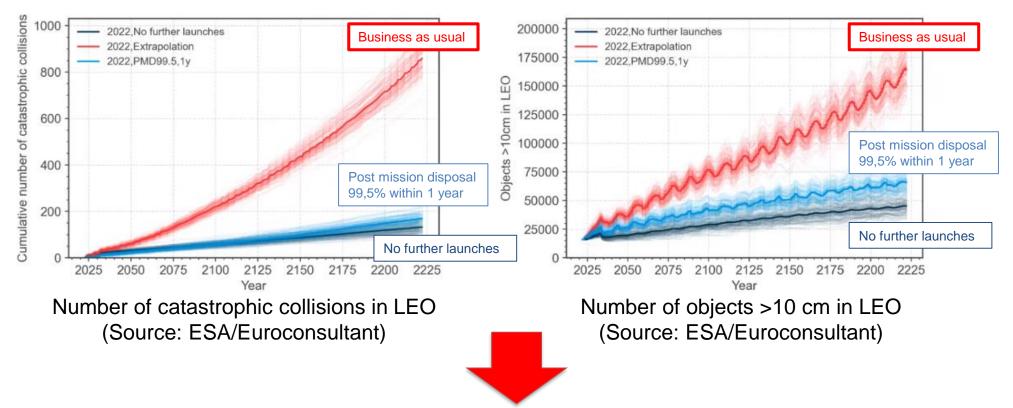
- ESA Projects for technology transfer:
 - Deployable Membranes (2014), ESA contract 4000112241/14/NL/SW
 - ADEO (2015), ESA contract 4000112253/14/NL/SW
 - ADEO-2 (2018), ESA contract 4000124984/18/NL/LvH-zk
 - AFO (2023), ESA contract 4000138835/22/NL/GLC/va





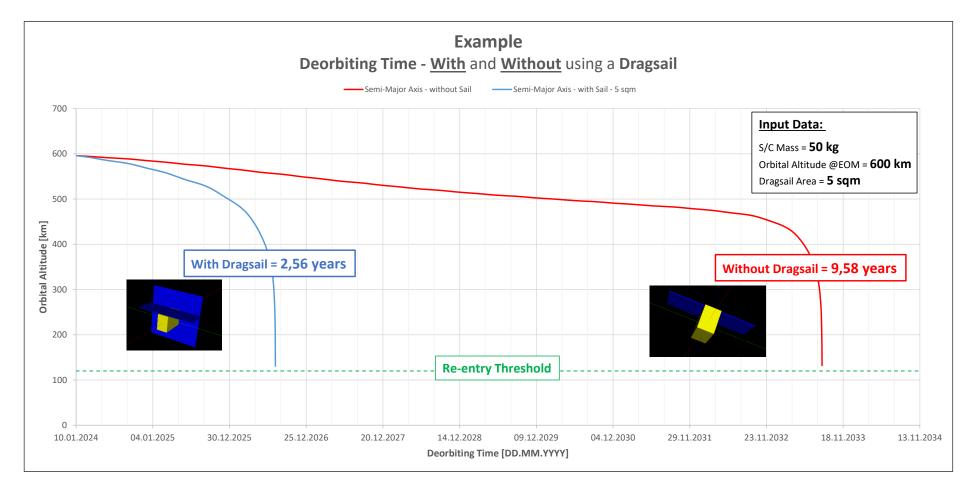
Orbital Population Evolution Models





New mission types and sustainable space activities need to be implemented to realise post mission disposals and to ensure future space missions – e.g. with missions using deorbit dragsails

Deorbit Example – S/C = 50 kg, Orbital Altitude = 600 km



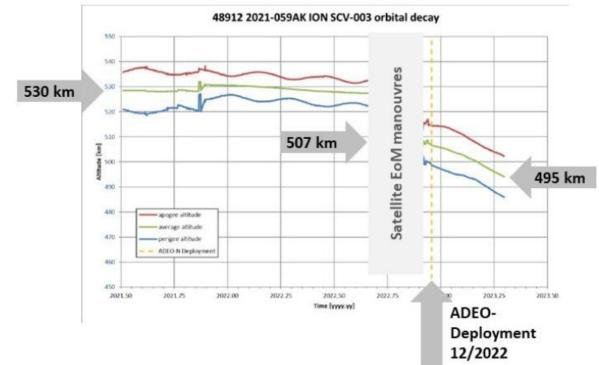








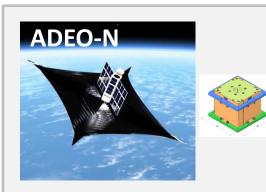
ADEO-N2 Heritage



- Deorbit without sail: 3 km within 10 months.
- Deorbit with sail: 12 km within 7 months (escalated: 17 km within 10 months).
- In general, the deorbit time with a deployed sail can be assumed to be 6 to 10 tmes quicker than without sail, with a mass penalty (ADEO-mass) of 1.0 kg (ADEO-N) to 3.5 kg (ADEO-M).



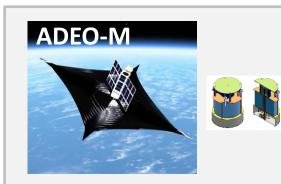
The ADEO Product Family



Application Satellites: 1 – 250 kg

Sail Sizes: nominal: 5 m² +/-2 m²

<u>Characteristics:</u> mass: 1 kg size: 10x10x10 cm

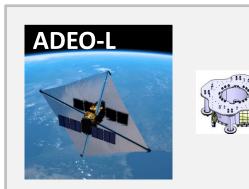


Application Satellites: 100 - 700 kg

Sail Sizes (tbc): nominal: 15 m² +/-5 m²

<u>Characteristics (tbc):</u> mass: 4 kg size: 20(dia.)x20 cm





Application Satellites: 200 -1,500 kg

<u>Sail Sizes:</u> nominal: 25 m² -5 / + 75 m²

<u>Characteristics:</u> mass: 9,5 kg size: 40x40x10 cm



ADEO Family



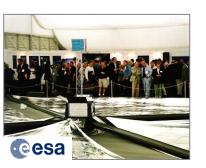
	ADEO-Cube	ADEO-N	ADEO-M	ADEO-L	
Status	In Development In Production beg. 2024	In Production	In Development In Production beg. 2024	In Production	
Technology Readiness Level (TRL)	4 (04/2023) 7 (12/2023)	9	4 (02/2022) 9 (06/2024)	7	
Satellite Masses [kg]	1-50	1-250	100-700	200-1.500	
ADEO Module Mass [kg]	0.5	0.8	4	9.5	
ADEO Module Size — stowed [cm x cm x cm]	9 x 9 x 5 (tbc)	10 x 10 x 10	20 x 20 x 20	40 x 40 x 10	
Sail Area [m ²]	2	5 ± 2	15 ± 5	25 +75/-5	
Deployment Mechanism	Mechanical	Mechanical	Mechanical	Motor	
Activation System	PyroCutter	PyroCutter	PyroCutter	Motor	
Mechanical Interface (*)	CubeSat Standard	4 x M5	4 x M5	8 x M6	
Electrical Interface (*)	2 Cables only	2 Cables only	2 Cables only	6 x RS-422	
Electrical Power (*)	12V @1A 1msec	12V @1A 1msec	12V @1A 1msec	24-38 V (<280 mW in Standby)	
Autonomous Functionality	Coming Soon	Coming Soon	Coming Soon	Available	



ADEO Launch- and Activity Log (1/3)







ADEO-L Proto -Development of the ADEO-L Ver.1 dragsail subsystem

2011



2014

Deployable Membrane Project in the frame of an ESA Contract -Extensive investigations on the dragsail membrane material ADEO-L1 -Development of the ADEO-L Ver.2 dragsail FM subsystem

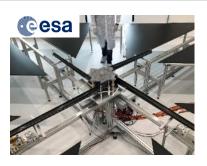
2018

..continuation, next page

Solar Sailing Activiy and Product Development in the frame of an ESA Contract - HPS was founded ©

2000







ADEO Launch- and Activity Log (2/3)

ADEO-N on







2018

ADEO-N on Electron Kickstage from RocketLab "It's Business Time" Mission



Parabolic Flight



ADEO-N on ION SCV003 from Dorbit, Launch with Falcon 9 from SpaceX

ADEO-N on ION SCV009 from Dorbit, Launch with Falcon 9 from SpaceX

✤ ADEO-N on EnduroSat ✤ ADEO-N on GenaSat

upcoming

2023

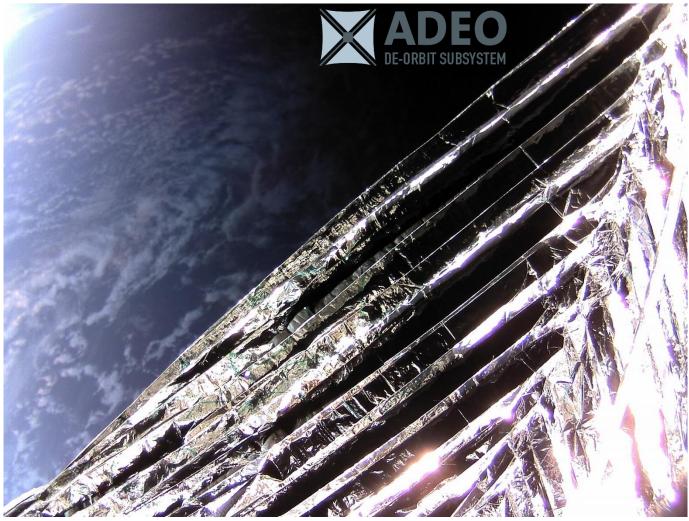


✤ ADEO-L on Redwire Space Mission



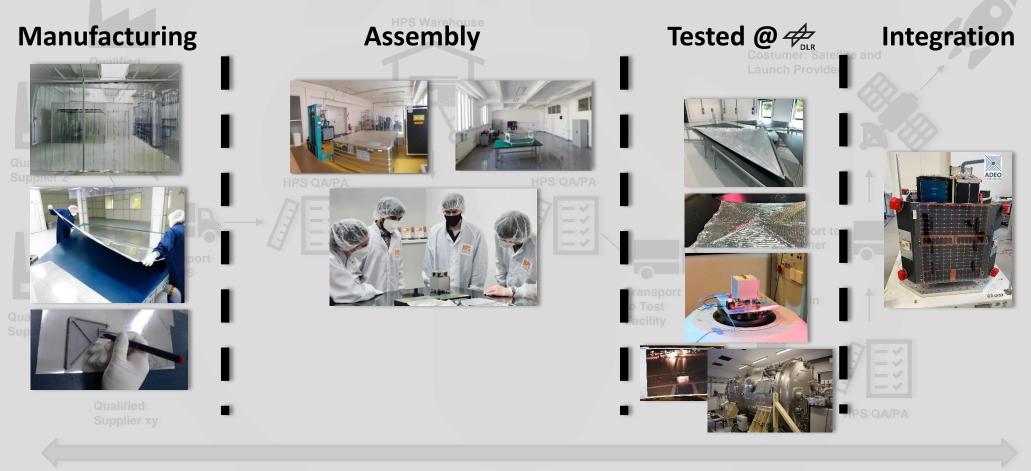
ADEO-N Deployment Verification in Orbit

Picture was captured on the **15th of December 2022** from ION the D-Orbit´s Satellite Carrier





The Route to Space: Manufacturing and Supply Chain



Risk Management

DEO

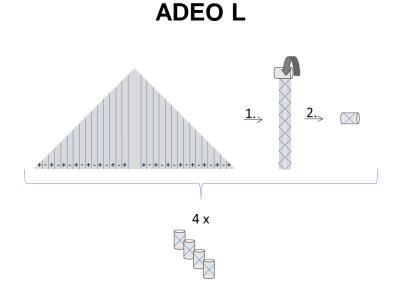
DE-ORBIT SUBSYSTEM

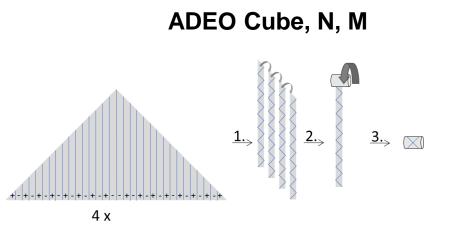


ADEO Deployment Strategy



- Different techniques were explored and implemented in resent years.
- Within the current AFO activity the ADEO products are streamlined, resulting in similar deploying strategies for all products.







ADEO Boom Technology

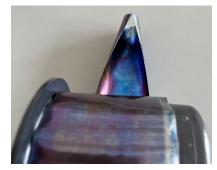
ADEO-Cube and ADEO-N bi-stable tape springs

✤ ADEO-M STEM

ADEO-L Composite Shell Booms (Gossamer-1)









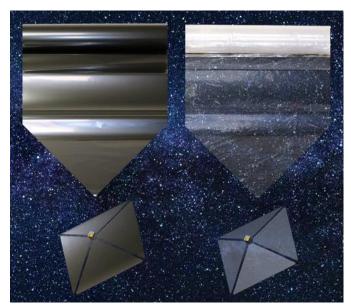


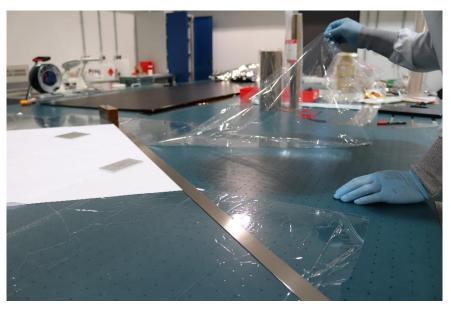


Non-Reflective Drag Sails



- NO to light pollution in the night sky!
- In cooperation with the DLR Institute of Space Systems, new non-reflective ATOX resistant materials are now implemented in the latest ADEO activity AFO.



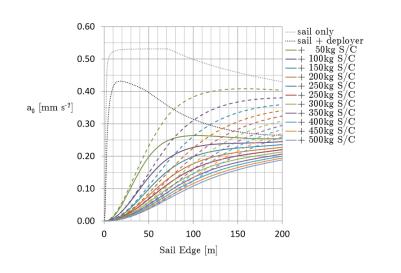




ADEO-L as a Solar Sail



- The ADEO-L technology is based on DLR's Gossamer-1 as described in Seefeldt et al. (2016) <u>https://elib.dlr.de/111077/</u>.
- An estimate of the characteristic acceleration for a system with a central non-ecjectable deployment mechanism has been derived in Seefeldt et al. (2021) <u>https://elib.dlr.de/146823/</u>:



- ▶ a_0 between 0.125 mm/s² and $0.2 \frac{\text{mm}}{\text{s}^2}$.
- Enabling first solar sail missions, e.g.
 - Multiple Near-Earth Asteroid
 Rendezvous or
 - Displaced Lagrange missions





CleanGreenSpace Missions

Deorbit dead satellites and expended launchers fast and reliably with our deployable dragsail:



More than 30,000 new on-orbit spacecraft endanger mission sustainability.

> ADEO products are suitable for satellites & launchers (1-700 kg) de-orbiting from LEO (< 800 km)

eesa

Let's Deorbit your Spacecraft Together!

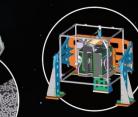






New Materials & Processes

Engineering & Integration Services

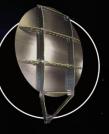




Thermal Hardware

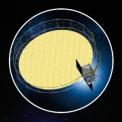


Lightweight Structures



Reflector Antennas





Deployable De-orbit Sails

Large Deploy. Reflector/ **Boom Subsystems**

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